

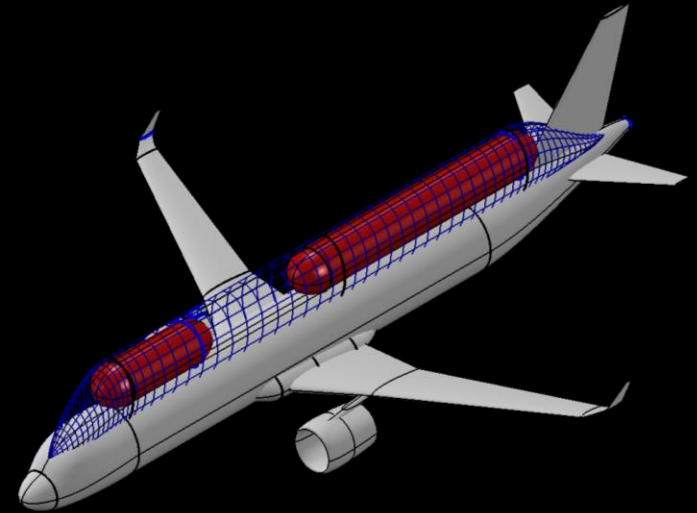
Hydrogen aircraft of the future: Technology, potential and performance



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Isak Jonsson and Christian Svensson

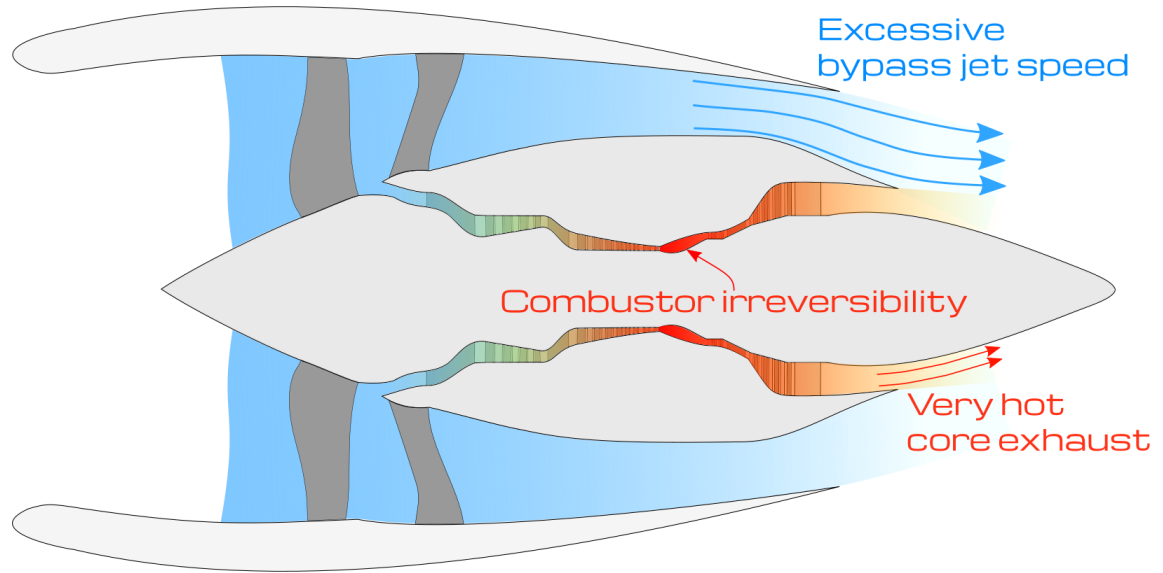
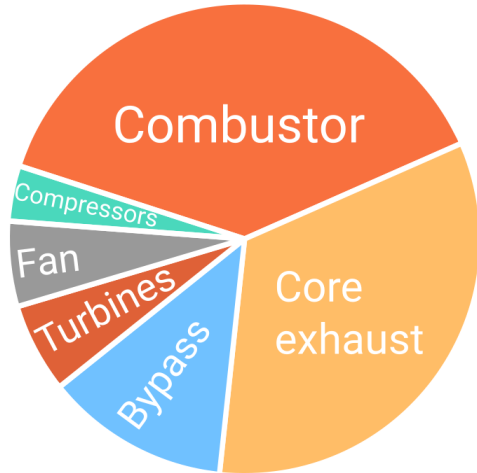


1. Can the flightpath 2050 targets be reached?



ULTIMATE (EU project - 2015 – 2018)

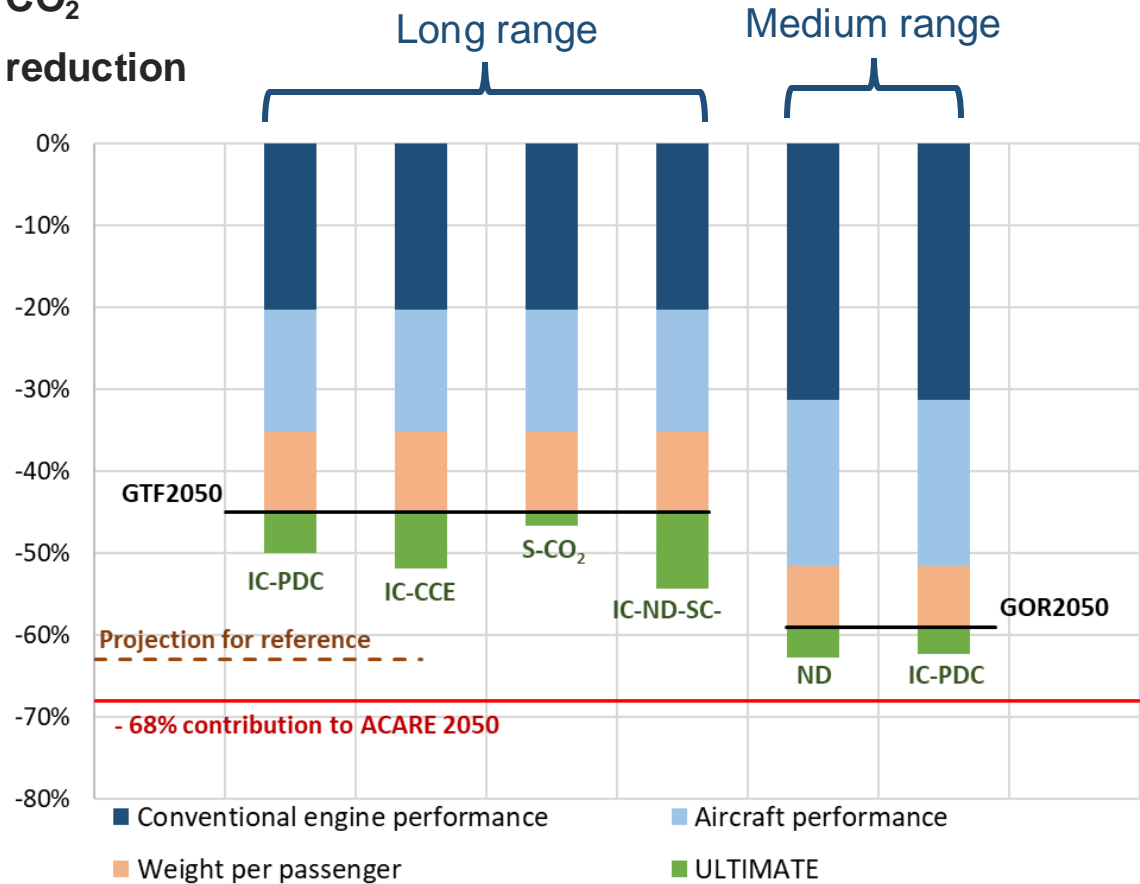
Ultra Low emission Technology Innovations for Mid-century Aircraft Turbine Engines



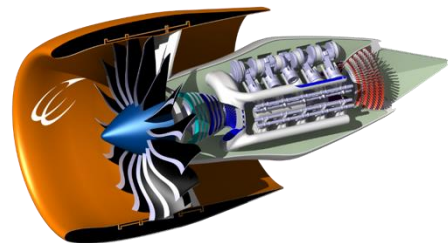
+ quite optimistic technology insertion for aircraft !

P. Heinemann, P. Panagiotou, Patrick Vratny, S. Kaiser, M. Hornung and K. Yakinthos, "Advanced Tube and Wing Aircraft for Year 2050 Timeframe," in 55th AIAA Aerospace Sciences Meeting , Grapevine, Texas, USA, 2017

CO₂ reduction



IC = Intercooled
PDC = Pulse detonation combustion
CCE = Composite Cycle Engine
ND = Nutating Disc



Can the targets be reached?
Probably not.

2. How far would the flightpath 2050 targets get us?



Climate impact of targets

How far would they get us?
Quite far actually!

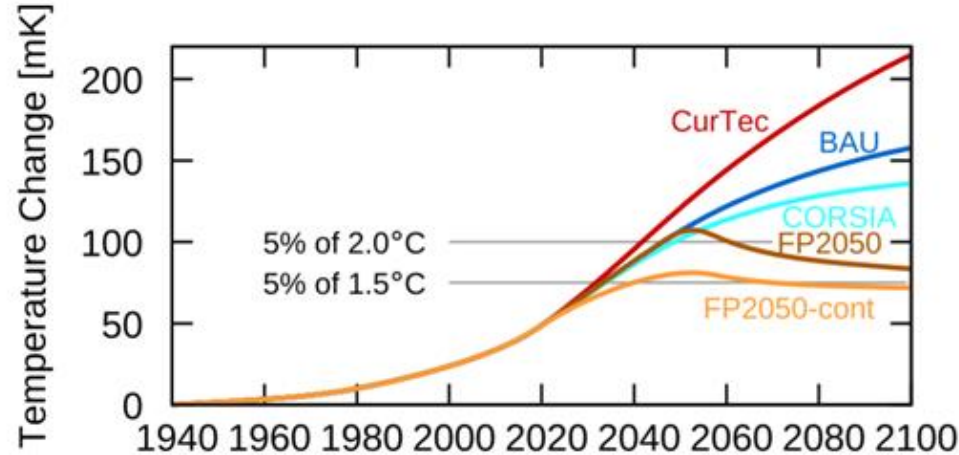


Fig. 2 Near-surface temperature change of five scenarios including CO₂ and non-CO₂-effects.

Grewe, V., Gangoli Rao, A., Grönstedt, T. *et al.* Evaluating the climate impact of aviation emission scenarios towards the Paris agreement including COVID-19 effects. *Nat Commun* **12**, 3841 (2021).

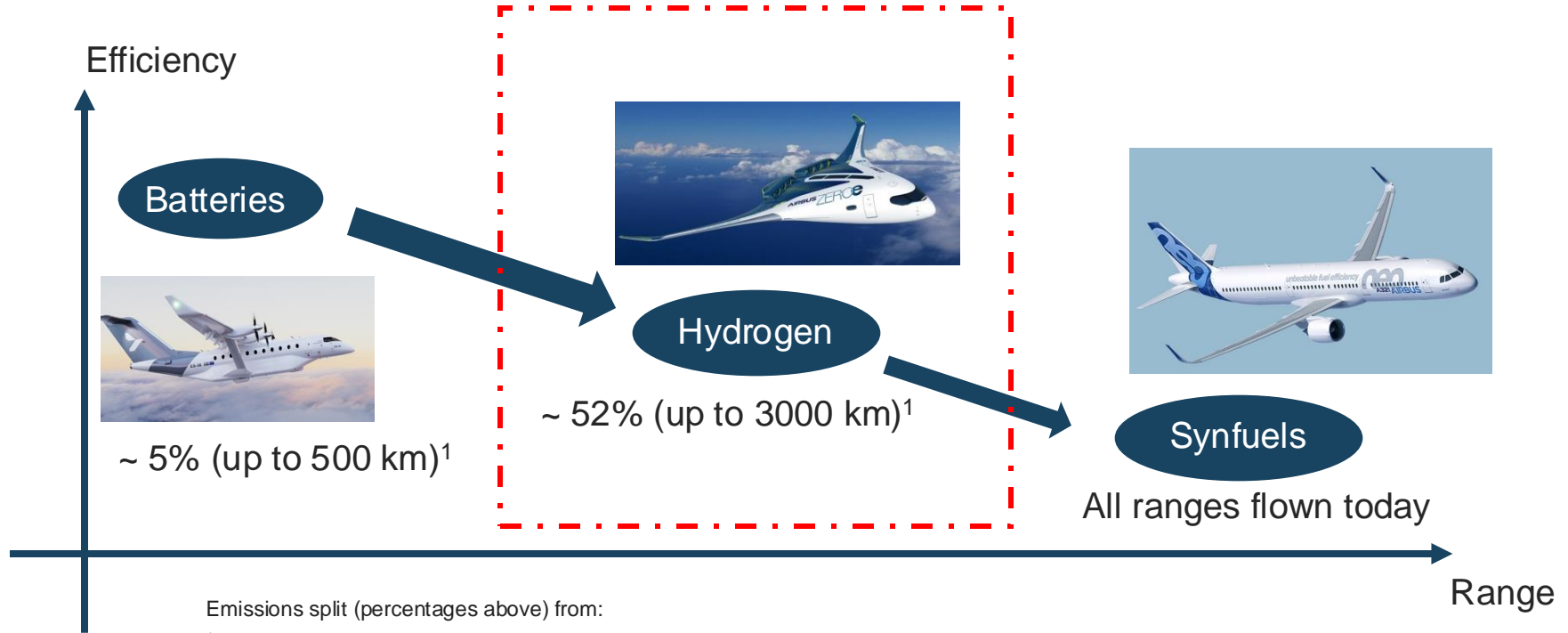
<https://doi.org/10.1038/s41467-021-24091-y>

1. Reachable - hardly
2. Enough - almost

We need new fuels!



Hydrogen can remove a large share of CO₂ emissions.



Emissions split (percentages above) from:

¹Graver, Brandon, Kevin Zhang, and Dan Rutherford. "emissions from commercial aviation, 2018." ICCT, 2019.

Rest of presentation will explore hydrogen aircraft:

1. Energy use vs range

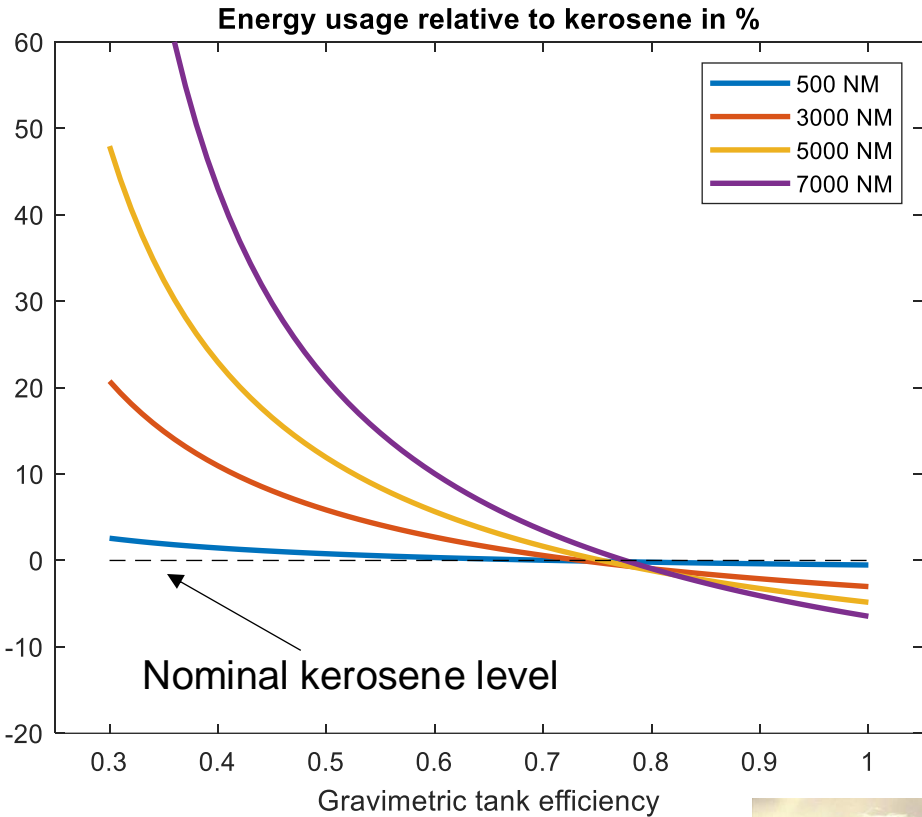
2. Propulsion heat management



Range – back of the envelope

- Energy need:
 - Breguet range for 777-200ER/GE90
 - Including added drag
 - Including added mass

- Now let's take a closer look at these aircraft

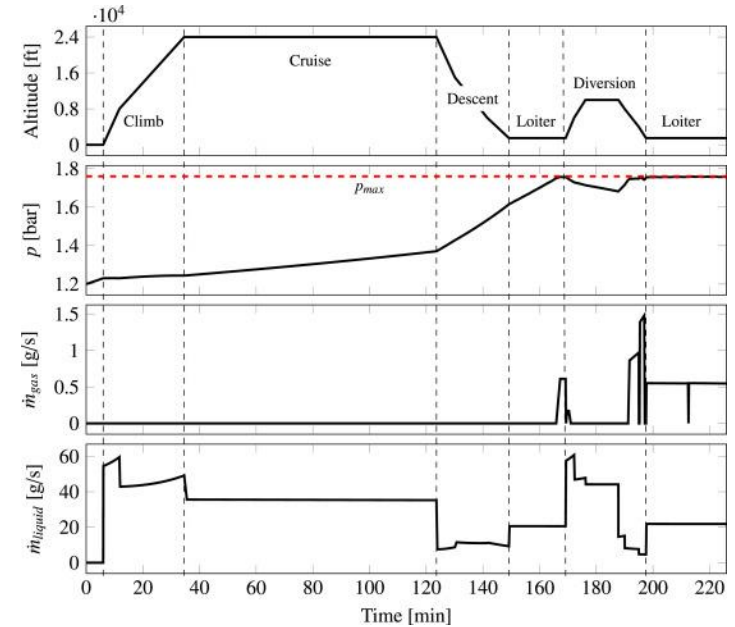
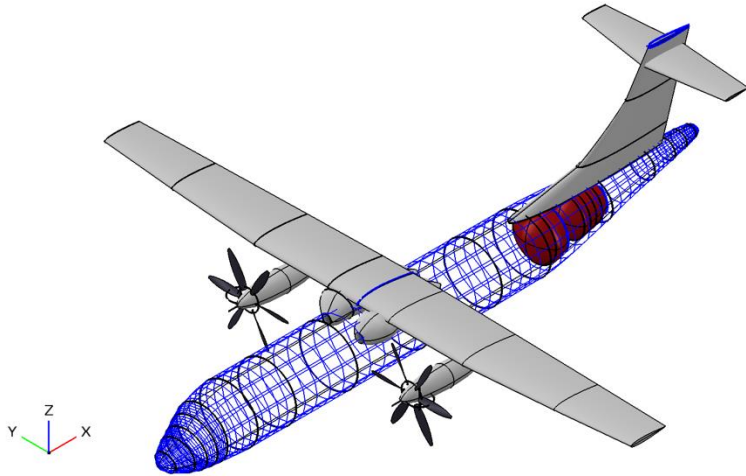


Shorter ranges

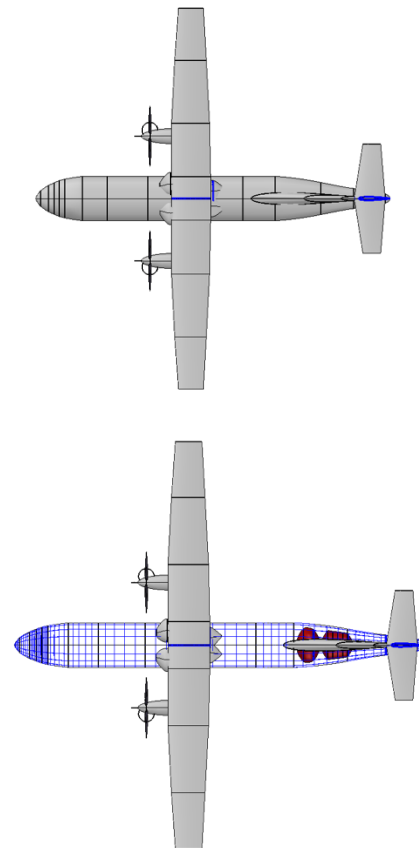


Air travel in the Nordic region

- State of the art fuel cell (PEM)
- Dynamic tank model
 - Optimization for design mission



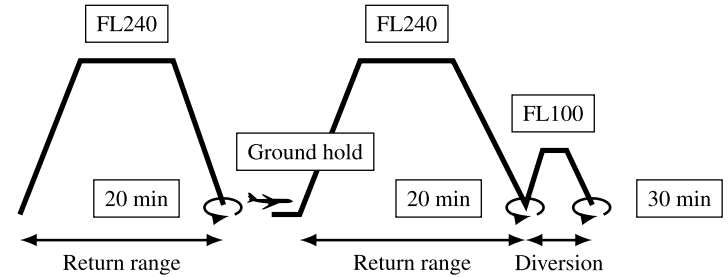
Parameter	ATR 42-500	LH2 PEMFC
MTOW	18.600 kg	21.049 kg
MZFW	17.000 kg	20.573 kg
Wing loading	341 kg/m ²	341 kg/m ²
Fuselage length	22.7 m	26.5 m
Fuselage fineness ratio	7.68	8.98
Fuselage wetted area*	208.2 m ²	243.6 m ²
Wing reference area	54.5 m ²	61.7 m ²
Wing aspect ratio*	10.9	10.9
Wing-span	24.57 m	26.16 m
Wing taper ratio*	0.56	0.56
Flap configuration	Single-slotted	Single-slotted
Lift over drag	16.254	16.480
Energy use		Approx. -25% (FC benefit). +11% if FC efficiency benefit is not included



Svensson, C., Oliveira, A. A., & Grönstedt, T. (2024).
 Hydrogen fuel cell aircraft for the Nordic market.
International Journal of Hydrogen Energy, 61, 650-663.

Return-without-refuel

- Close to 80% of theoretical range for a 12 hour on-ground scenario
- Heavier, well insulated tanks gives the most operational flexibility with a small loss in range
- We could operate whole Sweden with initially only three hydrogen hubs



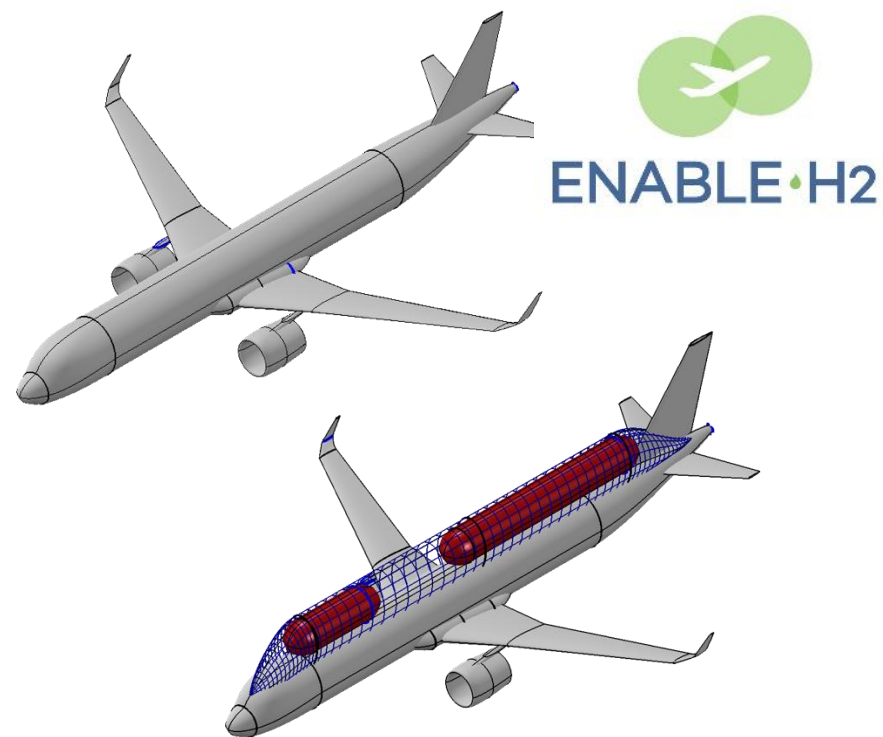
Medium range



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Parameter	Jet A-1 MR 2050	LH2 MR 2050
Mach	0.78	0.78
Maximum Takeoff Weight (MTOW) (kg)	77100	78600
Operational Empty Weight (OEW) (kg)	41400	51300 (+23.9%)
Design range (NM/km)	3000 NM / 5556 km	
Tank Volume (including insulation) (m ³)	-	35.4/80.2
Tank Weight (kg) (Incl. Structure/Fairing)	-	6505
Gravimetric efficiency	-	49.1% (67.6) ⁺
Mission fuel mass including reserves (kg)	14746	6280
Wing Loading (kg/m ²)	670	596
Wing Area (m ²)	117	133
Block Energy Use (MJ/PAX/km)	0.490	0.579 (+18%)
FPR (cruise/toc)	1.41/1.49	1.43/1.52
OPR (cruise/toc)	43.8/52.5	42.6/50.4
BPR (cruise/toc)	16.64/15.03	18.5/17
SFC (mg/Ns) (cruise/toc)	12.6	4.41/4.42

⁺Excluding fairing mass



- Increased wing loading not possible due to higher OEW for hydrogen aircraft

Long range



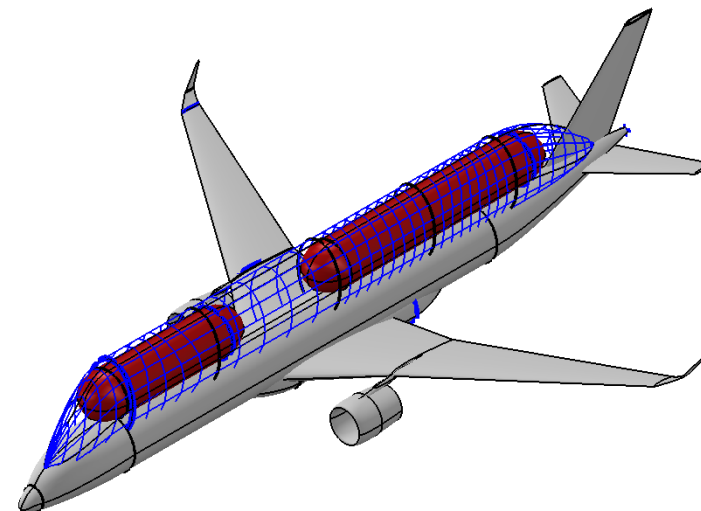
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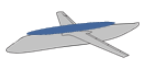





























ENABLE H2

Parameter	Jet A-1 LR 2050	LH2 LR 2050
Mach	0.85	0.85
Maximum Takeoff Weight (MTOW) (kg)	293700	284000
Operational Empty Weight (OEW) (kg)	152300	194300 (+27.6%)
Range (NM/km)		7500/13890
Tank Volume (including insulation) (m ³)	-	714
Tank Weight (kg) (Incl. Structure/Fairing)	-	26524
Gravimetric efficiency	-	61.4% (72.5%)+
Mission fuel mass including reserves (kg)		42100
Wing Loading (kg/m ²)	735	601.5
Wing Area (m ²)	410	477
Span (m)	69	70.5
Block Energy Use (MJ/PAX/km)	0.641 MJ/PAX	0.797 MJ/Pax (+24%)
FPR (cruise/toc)	1.42/1.47	1.39/1.45
OPR (cruise/toc)	68.2/75.3	67.4/76.5
BPR	19.6/19.2	23.1/21.7
SFC (mg/Ns) (cruise/toc)	12.24/12.51	4.34/4.35

+Excluding fairing mass

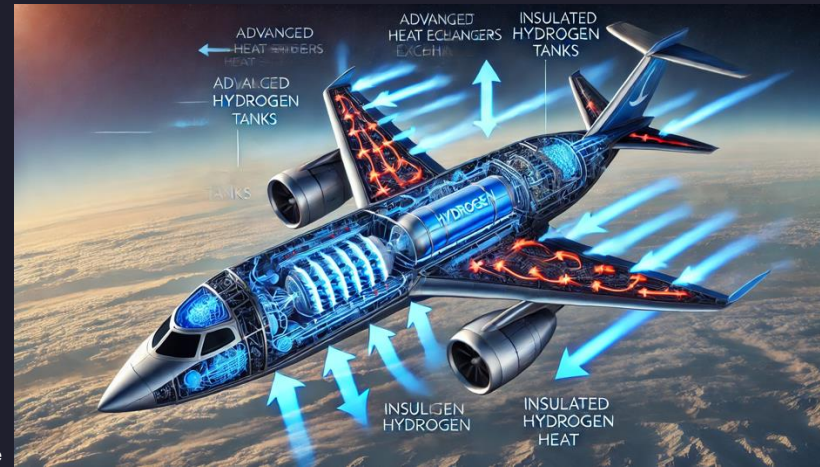


- All hydrogen aircraft ranges seem feasible
- Potentially large advantages for fuel cell aircraft if integration is successful
- Tank technology critical for longer ranges
- Synergy between advanced airframe designs and storage improves performance but is not necessary

 Baseline – low wings, tanks above fuselage	 1 – high wings, tanks above fuselage	 2 – low wings, external tanks below wings	 3 – high wings, external tanks below wings
 4 – low wings, external tanks above wings	 5 – external tanks joining box wings	 6 – conformal tanks either side of fuselage and below low wings	 7 – conformal tanks either side of fuselage and well below high wings
 8 – low wings, tank inside aft fuselage	 9 – high wings, tank inside aft fuselage	 10 – low wings, tanks in forward and aft fuselage	 11 – high wings, tanks in forward and aft fuselage
 12 – low wings, tank in centre of fuselage	 13 – high wings, tank in centre of fuselage	 14 – low wings, tank aft of double-bubble fuselage	 15 – high wings, tank aft of double-bubble fuselage
 16 – low wings, tanks at forward and aft ends of double-bubble fuselage	 17 – high wings, tanks at forward and aft ends of double-bubble fuselage	 18 – low wings, tank(s) in centre of double-bubble fuselage	 19 – high wings, tank(s) in centre of double-bubble fuselage
 20 – low wings, tanks aft and above double-deck fuselage	 21 – mid-height wings, tanks aft and above double-deck fuselage	 22 – low wings, tank(s) in centre of double-deck fuselage	 23 – mid-height wings, tank(s) in centre of double-deck fuselage
 24 – low wings, tanks at bottom of double-deck fuselage	 25 – mid-height wings, tanks at bottom of double-bubble fuselage	 26 – BWB with mid-height wings, internal under-floor tanks	 27 – BWB with mid-height wings, int. under-floor and external under-wing tanks
 28 – low wings, double fuselages, tanks inside aft fuselages	 29 – high wings, double fuselages, tanks inside aft fuselages	 30 – high wings, double fuselages, external tank on centreline	

Rompokos, P., Rolt, A., Nalianda, D., Isikveren, A. T., Senné, C., Gronstedt, T., & Abedi, H. (2021). Synergistic technology combinations for future commercial aircraft using liquid hydrogen. *Journal of Engineering for Gas Turbines and Power*, 143(7), 071017.

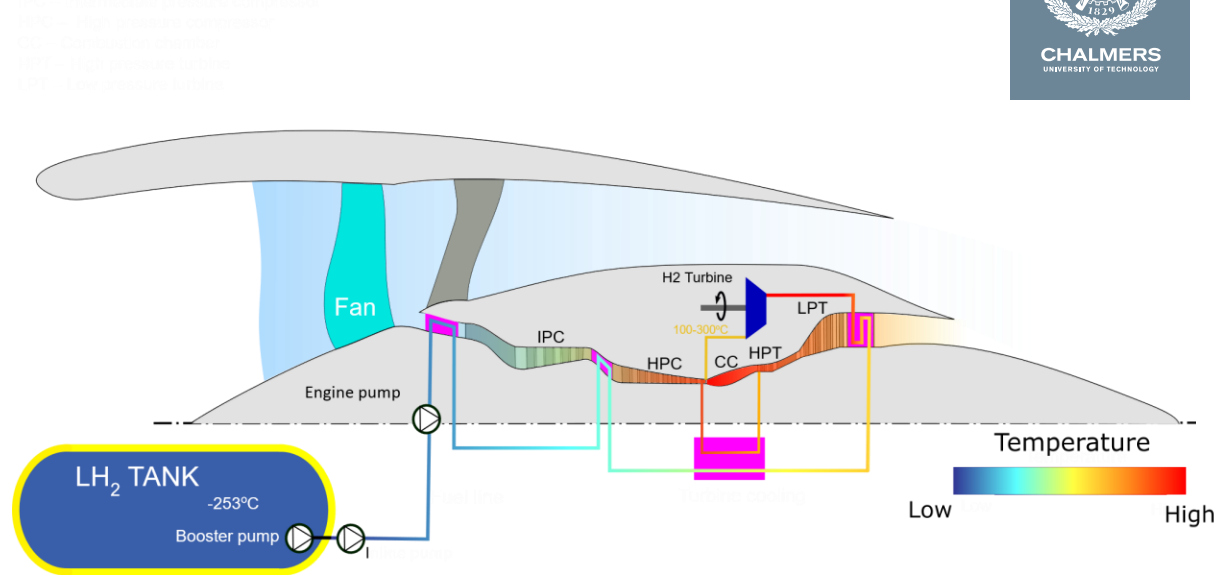
Heat management



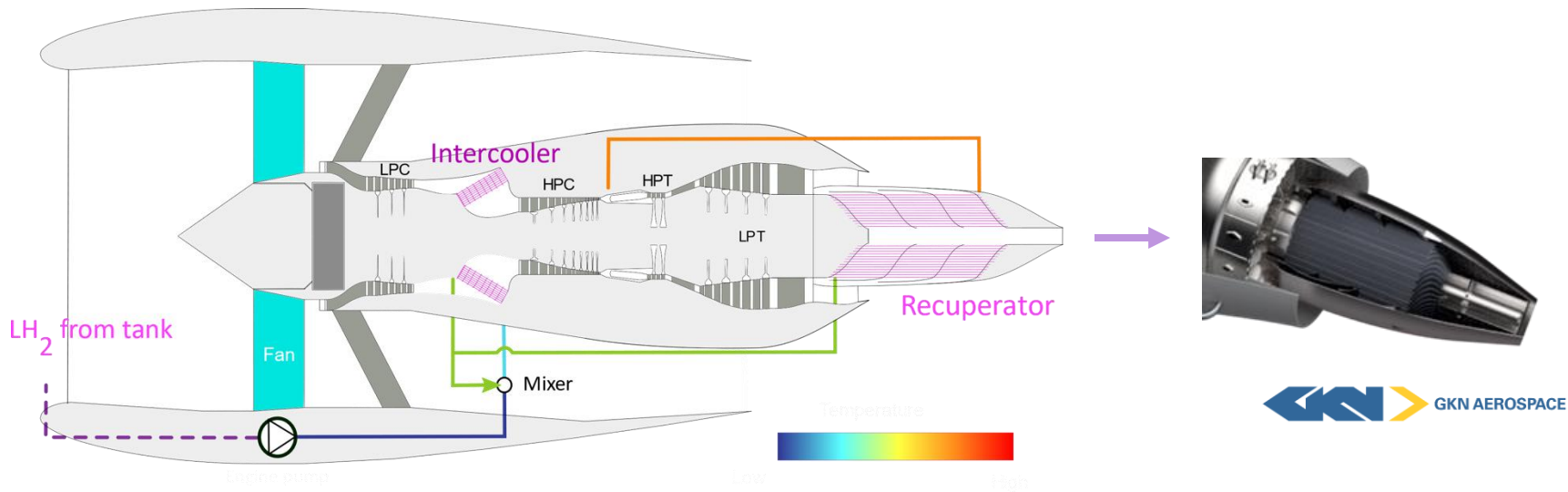
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Heat-management potential LH2

- Cryogenic storage
 - -253 C
- Hydrogen is exceptional coolant
 - Using existing surfaces
 - High heat capacity
 - More compact HEX



Temperature	Pressure	Enthalpy
22 K	2.3 bar	17.84 kJ/kg
700 K	40 bar	9793.5 kJ/kg
1000 K	40 bar	14229 kJ/kg



	Mass [kg]	Architecture	HEX mass* [kg]	Δ_{FB}
Baseline	3392	1+3+8+2+4	-	datum
Interc.	3402	1+3+6+2+4	129	-3.6%
Interc. – Recup.	3681	1+3+6+2+4	326	-5.5%

Patrao, Alexandre Capitaio, et al. "Compact heat exchangers for hydrogen-fueled aero engine intercooling and recuperation." *Applied Thermal Engineering* 243 (2024): 122538.

Better Planet | Air Travel | Engineering | Green Energy | Aviation | Sweden | Climate Change | Fossil Fuels

Short-Haul Flights May Be Fossil-Fuel-Free By 2045

Published Jul 19, 2024 at 10:21 AM EDT | Updated Jul 26, 2024 at 11:35 AM EDT

<https://www.newsweek.com/fossil-fuel-free-flights-hydrogen-power-air-travel-1924613>



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